## 가 LQG

## LQG Control for Semi-Active Suspension Systems with Road-Adaptation

(Hyun-Chul Sohn, Kyung-Tae Hong, and Keum-Shik Hong)

**Abstract** : A road-adaptive LQG control for the semi-active Macpherson strut suspension system of hydraulic type is investigated. A new control-oriented model, which incorporates the rotational motion of the unsprung mass, is used for control system design. First, based on the extended least squares estimation algorithm, a LQG controller adapting to the estimated road characteristics is designed. With computer simulations, the performance of the proposed LQG-controlled semi-active suspension is compared with that of a non-adaptive one. The results show better control performance of the proposed system over the compared one.

Keywords : semi-active suspension, new control-oriented model, road variation, extended least squares estimation algorithm, LQG controller





1.7Fig.1. A new model for the semi-active Macpherson suspension.



 2.
 1/4 LQG
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 Fig.
 2.
 Schematic of the road-adaptive LQG control for a 1/4-car model.



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 (Macpherson)
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 LQG
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$$(m_s + m_u)\ddot{z}_s + m_u l_C \cos(\theta - \theta_o)\ddot{\theta} - m_u l_C \sin(\theta - \theta_o)\dot{\theta}^2 + k_t \{z_s + l_C (\sin(\theta - \theta_o) - \sin(\theta_o)) - z_t\} = 0,$$
(1)

.

$$\begin{split} m_{u}l_{C}^{2}\ddot{\theta} + m_{u}l_{C}\cos(\theta - \theta_{o})\ddot{z}_{s} + k_{l}l_{C}\cos(\theta - \theta_{o})\{z_{s} + l_{C}(\sin(\theta - \theta_{o}) \\ -\sin(-\theta_{o})) - z_{r}\} - \frac{1}{2}k_{s}\sin(\alpha' - \theta)[b_{l} + d_{l}/\{c_{l} - d_{l}\cos(\alpha' - \theta)^{\frac{1}{2}}\}] \\ = -l_{B}f_{s}, \end{split}$$

$$(2)$$

$$\begin{split} a_{l} &= l_{A}^{2} + l_{B}^{2} , \ b_{l} = 2l_{A}l_{B} , \ c_{l} = a_{l}^{2} - a_{l}b_{l}\cos(\alpha') , \\ \alpha' &= \alpha + \theta_{o} , \ d_{l} = a_{l}b_{l} - b_{l}^{2}\cos(\alpha') \\ , & 1/4 \\ 1 & . \end{split}$$

1. 1/4 . Table 1. Nominal parameter values used in simulations.

-		
Parameters	Description	Nominal value
m <sub>s</sub>	Sprung mass	453 Kg
m <sub>u</sub>	Unsprung mass	36 Kg
k <sub>s</sub>	Coil spring constant	17,658 N/m
k <sub>t</sub>	Tire spring constant	183,887 N/m
$l_A$	Distance from O to A	0.66 m
$l_B$	Distance from O to B	0.34 m
$l_C$	Length of the control arm	0.37 m
α	Angle between the <i>y</i> -axis and $\overline{OA}$	74°
$\theta_0$	Angular displacement of the control arm at a static equilibrium point	$-2^{\circ}$

$$\begin{bmatrix} x_1 & x_2 & x_3 & x_4 \end{bmatrix}^T =$$

 $\begin{bmatrix} z_s & \dot{z}_s & \theta & \dot{\theta} \end{bmatrix}^T , \qquad y(t) = \ddot{z}_s(t) .$ 1/4
(1)-(2)  $x_e = (0, 0, \theta_o, 0)$ 

(1)-(2)

$$\dot{x}(t) = A_m x(t) + B_1 f_s + B_2 z_r(t), \ x(0) = x_o ,$$
(3)

$$y(t) = C_m x(t) + D_1 f_s + D_2 z_r , \qquad (4)$$

$$A_m = \begin{bmatrix} 0 & 1 & 0 & 0 \\ a_{21} & 0 & a_{23} & 0 \\ 0 & 0 & 0 & 1 \\ a_{41} & 0 & a_{43} & 0 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ -0.494 & 0 & 21.177 & 0 \\ 0 & 0 & 0 & 1 \\ -13,796 & 0 & -5105.4 & 0 \end{bmatrix},$$

(colored noise) (shaping filter)

,

$$S_{\nu}(\nu) = \frac{\sum_{j=0}^{m} b_{\nu j} \nu^{2j}}{\sum_{i=0}^{n} a_{\nu i} \nu^{2i}},$$
(5)

$$v$$
 ( )  
,  $a_{vi}$   $b_{vj}$  .

$$S_r(\omega)d\omega = S_v(v)dv , \qquad (6)$$

$$S_r(\omega)$$
 7  $\omega[rad/sec]$ ,  
 $\omega$  .

$$\omega = 2\pi v V . \tag{7}$$

.

$$dv = (2\pi V)^{-1} d\omega$$

$$S_{r}(\omega) = \frac{\sum_{j=0}^{m} b_{vj} (2\pi V)^{-2j-1} \omega^{2j}}{\sum_{i=0}^{n} a_{vi} (2\pi V)^{-2i} \omega^{2i}}.$$
(8)

7  
l
$$G_r(j\omega) = z_r(t)/\varepsilon(t) \qquad ,$$

[13].  

$$S_r(\omega) = \left| G_r(\omega) \right|^2 S_{\varepsilon}(\omega), \qquad (9)$$

$$S_{\varepsilon}(\omega) = S_r(\omega)$$

V

$$S_{\varepsilon}(\omega) = 1$$
 .

$$G_r(s) = \sqrt{(2\pi V)^{2n-2m-1}} \frac{\sqrt{b_{\nu m}} \prod_{j=0}^m (s-z_j)}{\sqrt{a_{\nu n}} \prod_{i=0}^n (s-p_i)}, \qquad (10)$$

$$s = j\omega$$
 ,  $p_i$ ,  $z_j$ 

.

$$\sum_{i=0}^{n} (-1)^{i} a_{vi} (2\pi V)^{-2i} s^{2i} = 0, \qquad (11)$$

$$\sum_{j=0}^{m} (-1)^{j} b_{\nu j} (2\pi V)^{-2j-1} s^{2j} = 0.$$
 (12)

m = 0, n = 2V (8)

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가 [2,27]. 2.

> 가 LQG

> > (13)

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θ

(10) ARMA  $\mathcal{E}(t)$ Z-

$$G_r(z^{-1}) = \frac{z_r(t)}{\varepsilon(t)} = \frac{b_{k_1}z^{-1} + b_{k_2}z^{-2}}{1 + a_{k_1}z^{-1} + a_{k_2}z^{-2}}.$$
 (13)

(regression model)

$$z_r(t) = \Phi^T(t-1)\theta ,$$

 $\Phi(t-1)$ 

Power spectrum density functions of the road input and shaping filter.

	Road profiles $S_r(\omega)$	Shaping filter $G_r(s)$
50km/h	0.7332	0.7332
Paved road	$\omega^4 - 191\omega^2 + 2.308 \times 10^5$	$s^2 + 27.75s + 480.4$
50km/hr	4.715	4.715
Standard unpaved road	$\overline{\omega^4 + 33000\omega^2 + 1.046 \times 10^8}$	$\overline{(s+59.14)(s+172.7)}$
40m/hr	15.37	15.37
Very poor unpaved road	$\overline{\omega^4 + 8633\omega^2 + 2.521 \times 10^6}$	$\overline{(s+17.10)(s+363.9)}$



$$\theta = [a_{k_1} \quad a_{k_2} \quad b_{k_1} \quad b_{k_2}]^T , \qquad (15)$$

9

 $\Phi(t-1) = [-z_r(t-1) \ -z_r(t-2) \ \varepsilon(t-1) \ \varepsilon(t-2)]^T.$ (16)

 $J_{LS}$ 

$$z_r(t)$$
  $\hat{z}_r(t)$ 

.

.

$$\hat{z}_r(t) = \boldsymbol{\Phi}^T(t-1)\hat{\boldsymbol{\theta}}(t) \ . \tag{17}$$

$$e_s(t) = z_r(t) - \hat{z}_r(t)$$
 7  
 $\hat{\theta}$ 

$$J_{LS}(\hat{\theta}, t) = \frac{1}{2} \sum_{i=1}^{t} \lambda^{t-i} \left\{ z_r(i) - \Phi^T(i-1)\hat{\theta}(i) \right\}^2, \quad (18)$$

$$\lambda (0 < \lambda \le 1) \qquad \text{(forgetting factor)} \quad ,$$
  
$$\lambda = 0.9 \qquad . \quad \Phi^T \Phi$$
  
$$J_{LS} \qquad \hat{\theta}(t) \qquad [29].$$

$$\mathcal{L}S$$
  $\mathcal{O}(l)$  [29].

$$\hat{\theta}(t) = \hat{\theta}(t-1) + L(t)[z_r(t) - \Phi^T(t)\hat{\theta}(t-1)], \qquad (19)$$

$$L(t) = \frac{P(t-1)\Phi(t)}{\lambda + \Phi^T(t)P(t-1)\Phi(t)},$$
(20)

$$P(t) = \frac{1}{\lambda} \{ I - L(t) \Phi^{T}(t) \} P(t-1) .$$
(21)

,

~

$$z_r \quad \varepsilon \qquad \ddot{z}_s \qquad .$$
  
 $z_r(s) \qquad \ddot{z}_s(s) \qquad (3)-(4)$ 

$$G_{rs}(s) \stackrel{\Delta}{=} \frac{\ddot{z}_s(s)}{z_r(s)} = C'(sI - A')^{-1}B_2 + D_2$$

$$= \frac{0.5s^4 + 17212.4s^3 + 317325s^2 - 11.5s - 212.6}{s^4 + 45.7s^3 + 509765s^2 + 17212.4s + 317359},$$
(22)

$$A' = A_m + B_1 H$$
 ,  $C' = C_m + D_1 H$  ,  
,  $B_2$  ,  $C_m$  ,  $D_2$  ) (3)-(4) .

7) 
$$f_s$$
 7)  
7)  $c_p \dot{\Delta} l \cong Hx \stackrel{\Delta}{=} c_p \left[ 0 \quad 0 \quad \partial \dot{\Delta} l / \partial \theta \quad \partial \dot{\Delta} l / \partial \dot{\theta} \right]_{x_e} x$ 

$$[9]. H = \begin{bmatrix} 0 & 0 & 0 & 614 \end{bmatrix}$$

$$z_r(s)$$
 (22)  
20Hz  
7} 7} .

$$z_r(s) = \frac{40\pi}{s + 40\pi} \cdot G^{-1}_{rs}(s) \cdot \ddot{z}_s(s) .$$
(23)

 $(A_m)$ 

(14)

(16) 
$$7^{\frac{1}{2}}$$
  
 $\varepsilon(t) \cong \zeta(t) = z_r(t) - \Phi^T(t-1)\hat{\theta}(t)$ . (24)  
 $\Phi(t-1)$   
 $\Phi(t-1) = [-z_r(t-1) - z_r(t-2) \zeta(t-1) \zeta(t-2)]^T$ . (25)  
3 50km/h ,  
. 3

0.3 . **IV. 7**†

가 가 가



4. . Fig. 4. Output feedback control system.

1. LQG

$$\begin{array}{ccc} 4 & 7 & f_s \\ y & W(z^{-1}) & z_r \\ \Phi_{ue} = M^* \Phi_{dd} S & \Phi_{dd} & (3)-(4) & 0.01 \\ & Z^- & . \end{array}$$

$$W(z^{-1}) = \frac{y(t)}{f_s(t)} \stackrel{\Delta}{=} \frac{B(z^{-1})}{A(z^{-1})},$$
(26)

$$W_d(z^{-1}) = \frac{y(t)}{z_r(t)} \stackrel{\Delta}{=} \frac{C_d(z^{-1})}{A(z^{-1})} .$$
(27)

$$S(z^{-1}) = \frac{y(t)}{d(t)} = \frac{1}{1 + W(z^{-1})K(z^{-1})},$$
(28)

$$M(z^{-1}) = -\frac{f_s(t)}{d(t)} = \frac{K(z^{-1})}{1 + W(z^{-1})K(z^{-1})}.$$
(29)

[28].

LQG

$$J = \frac{1}{2\pi j} \oint_{|z|=1} X(z^{-1}) \frac{dz}{z}$$

$$= \frac{1}{2\pi j} \oint_{|z|=1} \left\{ Q_c \Phi_{ee} + R_c \Phi_{uu} + G_c \Phi_{ue} + G_c^* \Phi_{eu} \right\} \frac{dz}{z}.$$
(30)
$$* \quad (adjoint) \quad , \ Q_c(z^{-1}) , \ R_c(z^{-1}) , \ G_c(z^{-1})$$

$$7$$

$$Q_{c} = \frac{B_{q}^{*}B_{q}}{A_{q}^{*}A_{q}}, \ R_{c} = \frac{B_{r}^{*}B_{r}}{A_{r}^{*}A_{r}}, \ G_{c} = \frac{B_{q}^{*}B_{r}}{A_{q}^{*}A_{r}}.$$
 (31)

$$\Phi_{uu} = M \Phi_{dd} M^*, \qquad (32)$$

$$\Phi_{ee} = S \Phi_{dd} S^* , \qquad (33)$$

$$\Phi_{ue} = M^* \Phi_{dd} S , \qquad (34)$$

 $X(z^{-1})$ 

 $\Phi_{dd}$ 

$$\Phi_{dd} = Y_f^* Y_f = \left\{ \frac{C_d}{A} \varepsilon(t) \right\}^* \left\{ \frac{C_d}{A} \varepsilon(t) \right\} = \frac{C_d^* C_d}{A^* A} .$$
(35)

$$X(z^{-1}) = Y_{f}^{*}[M^{*}(W^{*}Q_{c}W + R_{c} - W^{*}G_{c} - G_{c}^{*}W)M + Q_{c} - M^{*}W^{*}Q_{c} - Q_{c}WM + MG_{c} + M^{*}G_{c}^{*}]Y_{f}.$$
(36)

$$(36) Y_c \Phi_h$$

$$Y_c^* Y_c = W^* Q_c W + R_c - W^* G_c - G_c^* W , \qquad (37)$$

9

$$\Phi_h = W^* \Phi_{dd} Q_c - G_c^* \Phi_{dd} .$$
(38)

 $X(z^{-1})$  (37) (38) (36)

$$X = Y_{f}^{*}[M^{*}(W^{*}Q_{c}W + R_{c} - W^{*}G_{c} - G_{c}^{*}W)M + Q_{c} - M^{*}W^{*}Q_{c} - Q_{c}WM + MG_{c} + M^{*}G_{c}^{*}]Y_{f} = (Y_{c}MY_{f} - Y_{c}^{*-1}\Phi_{h}Y_{f}^{*-1})^{*}(Y_{c}MY_{f} - Y_{c}^{*-1}\Phi_{h}Y_{f}^{*-1}) + Q_{c}\Phi_{dd} - Y_{c}^{*-1}\Phi_{h}Y_{f}^{*-1}Y_{f}^{-1}\Phi_{h}^{*}Y_{c}^{-1} .$$
(39)

2.

(39) 2 
$$M(z^{-1})$$
  
. (37) (26) (31)  
.

$$Y_{c}^{*}Y_{c} = \frac{D_{c}^{*}D_{c}}{AA^{*}A_{q}A_{q}^{*}A_{r}A_{r}^{*}}.$$
(40)

(40)

$$Y_c M Y_f = \frac{D_c C_d K_n}{A A_q A_r (A K_d + B K_n)},$$
(41)

$$Y_c^{*-1} \Phi_h Y_f^{*-1} = \frac{B_q C_d \left(A_r^* B^* B_q^* - A^* A_q^* B_r^*\right)}{D_c^* A A_q} \,. \tag{42}$$

Diophantine

(43)

$$\begin{array}{ccc} . & (42) & D_c^* & AA_q \\ G & F & Diophantine \end{array}$$

$$D_c^*Gz^{-g} + FAA_a = B_aC_d(A_r^*B^*B_a^* - AA_a^*B_r^*)z^{-g},$$

(39)

$$g$$
 (43)  $z^{-1}$ 

$$z^{-1}$$
 7  $(39)$ 

.

$$Y_{c}MY_{f} - Y_{c}^{*-1}\Phi_{h}Y_{f}^{*-1} = \frac{D_{c}C_{d}K_{n}}{AA_{q}A_{r}(AK_{d} + BK_{n})} - \frac{G}{AA_{q}} - \frac{Fz^{g}}{D_{c}^{*}}.$$
(44)

(44)  $D_c C_d$  Diophantine . Diophantine H F, F (43) .

 $D_{c}^{*}A_{r}Hz^{-g} - FBA_{r}A_{q} = (B_{r}B_{r}^{*}A^{*}A_{q}A_{q}^{*} - B^{*}B_{q}^{*}B_{r}A_{q}A_{r}^{*})C_{d}z^{-g}.$ (45)

3. 
$$7 \downarrow W_h(f)$$
 (BS6841).

Table 3. Frequency characteristics of weighting functions  $W_h(f)$  to assess human exposure to whole-body vibration (BS6841).

Exposure	Measure	Weighting	Multiplying	Eracijonav rasponsa
area	axis	function	factor	Frequency response
Seat	x <sub>seat</sub> , y <sub>seat</sub>	w <sub>d</sub>	1.00	$\begin{array}{l} 0.5 < f < 2.0 \\ : W_h(f) = 1.0 \\ 2.0 < f < 80.0 \\ : W_h(f) = 2.0 / f \end{array}$
	z <sub>seat</sub>	w <sub>b</sub>	1.00	$\begin{array}{l} 0.5 < f < 2.0 \\ : W_h(f) = 0.4 \\ 2.0 < f < 5.0 \\ : W_h(f) = f / 5.0 \\ 5.0 < f < 16.0 \\ : W_h(f) = 1.0 \\ 16.0 < f < 80.0 \\ : W_h(f) = 16/f \end{array}$
	$R_x$	w <sub>e</sub>	0.63	$\begin{array}{l} 0.5 < f < 1.0 \\ : W_h(f) = 0.63 \\ 1.0 < f < 5.0 \\ : W_h(f) = 0.63 / f \end{array}$
	R <sub>y</sub>	W <sub>e</sub>	0.40	$\begin{array}{l} 0.5 < f < 1.0 \\ : W_h(f) = 0.4 \\ 1.0 < f < 5.0 \\ : W_h(f) = 0.4 / f \end{array}$
	R <sub>z</sub>	W <sub>e</sub>	0.20	$\begin{array}{l} 0.5 < f < 1.0 \\ : W_h(f) = 0.2 \\ 1.0 < f < 5.0 \\ : W_h(f) = 0.2 / f \end{array}$
Back	x <sub>b</sub>	W <sub>c</sub>	0.80	$\begin{array}{l} 0.5 < f < 8.0 \\ : W_h(f) = 0.8 \\ 8.0 < f < 80.0 \\ : W_h(f) = 6.4 / f \end{array}$
	y <sub>b</sub>	w <sub>d</sub>	0.50	$\begin{array}{l} 0.5 < f < 2.0 \\ : W_h(f) = 0.5 \\ 2.0 < f < 80.0 \\ : W_h(f) = 1.0 / f \end{array}$
	z <sub>b</sub>	w <sub>d</sub>	0.40	$\begin{array}{l} 0.5 < f < 2.0 \\ : W_h(f) = 0.4 \\ 2.0 < f < 80.0 \\ : W_h(f) = 0.8 / f \end{array}$
Feet	$x_f$ , $y_f$	w <sub>b</sub>	0.25	$\begin{array}{l} 0.5 < f < 2.0 \\ : W_h(f) = 0.1 \\ 2.0 < f < 5.0 \\ : W_h(f) = f/20.0 \\ 5.0 < f < 16.0 \\ : W_h(f) = 0.25 \\ 16.0 < f < 80.0 \\ : W_h(f) = 4.0/f \end{array}$
	$z_f$	w <sub>b</sub>	0.40	$\begin{array}{c} 0.5 < f < \overline{2.0} \\ \vdots W_h(f) = 0.16 \\ 2.0 < f < 5.0 \\ \vdots W_h(f) = f / 12.5 \\ 5.0 < f < 16.0 \\ \vdots W_h(f) = 0.4 \\ 16.0 < f < 80.0 \\ \vdots W_h(f) = 6.4 / f \end{array}$

$$D_{c}^{*}(GA_{r}B + HAA_{r})z^{-g} = D_{c}^{*}D_{c}C_{d}z^{-g}$$
. (46)

.

(46) 
$$D_c^* z^{-g} D_c C_d$$
  
.  
 $GA_r B + HAA_r = D_c C_d$ . (47)

(47) (44)

$$Y_{c}MY_{f} - Y_{c}^{*-1}\Phi_{h}Y_{f}^{*-1} = \frac{HK_{n} - GK_{d}}{A_{q}(AK_{d} + BK_{n})} - \frac{Fz^{g}}{D_{c}^{*}}.$$
 (48)

(48) (causal) (non-causal)

K

$$K = \frac{K_n}{K_d} = \frac{G}{H} \,. \tag{49}$$

가

$$u_s(t) = -Ky(t) . (50)$$

3. 가

> LQG (31) 가

7) 
$$Q_c(z^{-1}), R_c(z^{-1}), G_c(z^{-1})$$
  
7)

2 ISVR Griffin [17,19] 가 가가 (equivalent weighting filter) 3 가 (BS6841) [18,20], 5

$$T(s) = \frac{277s + 29293}{s^2 + 182.8s + 10106} - \frac{19s + 1163}{s^2 + 26.5s + 632} .$$
 (51)

가

(13)

•

가

$$T(z^{-1}) = \frac{B_f(z^{-1})}{A_f(z^{-1})} = \frac{1 + b_{f_1}(z^{-1}) + b_{f_2}(z^{-1}) + b_{f_3}(z^{-1}) + b_{f_4}(z^{-1})}{1 + a_{f_1}(z^{-1}) + a_{f_2}(z^{-1}) + a_{f_3}(z^{-1}) + a_{f_4}(z^{-1})}.$$
(52)

$$Q_{c}(z^{-1}) = \frac{B_{q}^{*}(z^{-1})B_{q}(z^{-1})}{A_{q}^{*}(z^{-1})A_{q}(z^{-1})} = \frac{B_{f}^{*}(z^{-1})B_{k}^{*}(z^{-1})B_{k}(z^{-1})B_{f}(z^{-1})}{A_{f}^{*}(z^{-1})A_{k}^{*}(z^{-1})A_{k}(z^{-1})A_{f}(z^{-1})}$$
(53)
$$R_{c}(z^{-1}) = \frac{B_{r}^{*}(z^{-1})B_{r}(z^{-1})}{x^{*}} = \frac{\rho^{2}B_{f}^{*}(z^{-1})B_{k}^{*}(z^{-1})B_{k}(z^{-1})B_{f}(z^{-1})}{x^{*}},$$

$$= \frac{1}{A_r^*(z^{-1})A_r(z^{-1})} = \frac{1}{A_f^*(z^{-1})A_k^*(z^{-1})A_k(z^{-1})A_f(z^{-1})},$$
(54)



Fig. 5. Frequency characteristics of equivalent filters to weighting functions to assess human response to vibration.



Fig. 6. Damping force characteristics of a typical continuously variable damper.

$$G_{c}(z^{-1}) = \frac{B_{q}^{*}(z^{-1})B_{r}(z^{-1})}{A_{q}^{*}(z^{-1})A_{r}(z^{-1})} = \frac{\rho B_{f}^{*}(z^{-1})B_{k}^{*}(z^{-1})B_{k}(z^{-1})B_{f}(z^{-1})}{A_{f}^{*}(z^{-1})A_{k}^{*}(z^{-1})A_{k}(z^{-1})A_{f}(z^{-1})}.$$
 (55)

가

가

4.

$$f_{s} = \begin{cases} f_{s}^{*}, & if \qquad f_{s}^{*} \le u_{s}, \\ u_{s}, & if \qquad f_{s}^{*} < u_{s} < f_{s}^{*}, \\ f_{s}^{*}, & if \qquad f_{s}^{*} \ge u_{s}, \end{cases}$$
(56)

가

가

가 가 0A



7. Lookup table

















(b) Conventional and road adaptive control of semi-active suspension.

15

Time [sec]

20

25

30

 9.
 71
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 Fig.
 9. Acceleration responses of the sprung mass.

Road-Adaptiv



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